RSC Advances

This is an *Accepted Manuscript*, which has been through the Royal Society of Chemistry peer review process and has been accepted for publication.

Accepted Manuscripts are published online shortly after acceptance, before technical editing, formatting and proof reading. Using this free service, authors can make their results available to the community, in citable form, before we publish the edited article. This *Accepted Manuscript* will be replaced by the edited, formatted and paginated article as soon as this is available.

You can find more information about *Accepted Manuscripts* in the Information for Authors.

Please note that technical editing may introduce minor changes to the text and/or graphics, which may alter content. The journal's standard Terms & Conditions and the Ethical quidelines still apply. In no event shall the Royal Society of Chemistry be held responsible for any errors or omissions in this *Accepted Manuscript* or any consequences arising from the use of any information it contains.

www.rsc.org/advances

Exploring highly energetic aliphatic azido nitramines for plasticizers

Junqing Yang¹, Xueli Zhang¹, Pin Gao², Xuedong Gong^{1*}, Guixiang Wang^{1*}

1. Department of Chemistry, Nanjing University of Science and Technology, Nanjing 210094, China

2. National Civil Blasting Equipment Quality Supervision and Testing Center, Nanjing 210094, China

Abstract:

A set of energetic aliphatic azido nitramines was designed and studied with the B3LYP/6-31G* method of density functional theory to find new promising azido nitramines applicable to propellants as plasticizers and to investigate the effect of the methylene nitramino group (CH_2NNO_2) on performance. The reliability of the theoretical method was tested by comparing the theoretical densities with the experimental data. The thermodynamic properties (heat capacity, enthalpy, entropy, and heat of formation), density (*ρ*), detonation pressure (P) , detonation velocity (D) , specific impulse (I_s) and stability were predicted. With the increase in the number of -CH₂NNO₂- groups (n) , the thermodynamic properties, ρ , P , and *D* increase, while the stability decreases slightly. There are linear relationships between thermodynamic properties and n . A -CH₂NNO₂- group improves the heat of formation by 67.59 kJ.mol⁻¹ on average. *I_s* reaches to the maximum when $n=4$ and changes little when $n\geq 5$. It is worth noting that *I*ss of all title compounds are higher than that of HMX (1, 3, 5, 7-tetranitro-1, 3, 5, 7-tetraazacyclooctane, a famous explosive).

Key words: aliphatic azido nitramine, energetic plasticizer, specific impulse, stability

1 Introduction

The common way to improve the output energy of propellants is to add certain amounts of high energetic explosives, such as, RDX (hexahydro-1, 3, 5-trinitro-1, 3, 5-trizine) and HMX (1, 3, 5, 7-tetranitro-1, 3, 5, 7- tetraazacyclooctane). However, this may also bring some problems, such as, high flame temperature and serious ablation of gun barrel. Therefore, it is necessary to mix energetic plasticizers or other additives into the propellants. Nitrate ester plasticizers have good plasticizing effects and high energy, but they also have undesired high

i,

^{*} Corresponding authors. E-mail: gongxd325@mail.njust.edu.cn; wanggx1028@163.com

Tel: +86-25-84315947-803

RSC Advances Page 2 of 15

volatility and impact sensitivity. Compared with nitrate ester plasticizers, azido plasticizers, especially azido nitramine plasticizers which have two kinds of energetic groups (i.e., nitramino (-NNO₂) and azido (-N₃)) in molecules have better stability and compatibility with other components of propellants (e.g. binders) $\frac{1}{1}$ and have been attracting more and more attentions $2-10$. Researches found that azido nitramine plasticizers can adjust the average molecular weight of combustion products, increase the nitrogen content and hence reduce the characteristic signal, and improve the energy and specific impulse of propellants. Although this type of compounds has attracted great interests, effects of structures on their performance are still unclear. Therefore, in this paper, eight aliphatic azido nitramines (Fig. 1) were studied using the density functional theory (DFT). Thereinto, compounds $1-4$ have been synthesized and reported $2-4$, particularly for compound 3 (1, 7-diazido-2, 4, 6-trinitrazaheptane), extensive studies have been carried out both experimentally and theoretically $8-11$.

Since the output energy (detonation pressure, detonation velocity, specific impulse) and stability are the important parameters for characterizing and evaluating the performance of energetic materials, they, together with the related thermodynamic properties and density of these compounds were investigated. In terms of the energetic property and stability, we try to (I) search for new azido nitramine plasticizers with better properties and (II) investigate the role of $-CH_2NNO_2$ - groups in the molecular design of the azido nitramines. Since specific impulse (I_s) is the most vital parameter to weigh the performance of propellants, in this work, two kinds of idealized stoichiometric decomposition reactions were designed to predict reliable *I*ss. Based on the well-predicted *I*ss to screen promising energetic plasticizers makes this study more meaningful.

Fig. 1 Structures of title compounds with the numbering of some $N-NO₂$ bonds

2 Theoretical methods

Molecular geometries of the title compounds were fully optimized using the DFT-B3LYP method $12-13$ in combination with the 6-31G* basis set 14 . Many studies $15-16$ have shown that accurate molecular structures and other properties can be produced at this level. The density (ρ) was evaluated with $\rho = M/V_m$, where *M* is the molecular weight. *V*_m is the average molar volume obtained with the Monte Carlo integration scheme by setting the number of points to 20000 per Bohr³, which is basically equivalent to the average of 1000 default runs ¹⁷. Previous studies ¹⁷⁻¹⁸ have indicated that the theoretical ρ calculated using this method is very close to the experimental crystal density for azido compounds.

To discuss the effect of the number (n) of $-CH₂NNO₂$ - groups on the thermodynamic properties, heat capacity $(C^0_{p,m})$, enthalpy (S^0_m) , and entropy (H^0_m) at different temperatures (200 to 800 K) were calculated using a self-compiled program and the heat of formation (HOF) was calculated from the isodesmic reaction (1). This procedure has been proved to be very successful 19 and widely used in previous studies $^{20-22}$.

$$
N_3(CH_2)_{n+1}(NNO_2)_nN_3 + (n+1)CH_4 + 2nNH_3 \longrightarrow 2CH_3N_3 + nNH_2NO_2 + 2nCH_3NH_2 (n=1 \sim 8)
$$
 (1)

The reaction enthalpy (ΔH_r) of the isodesmic reaction at 298 K was calculated using the following equation:

RSC Advances **Page 4 of 15**

$$
\Delta H_{\rm r} = \Sigma \Delta H_{\rm f,P} - \Sigma \Delta H_{\rm f,R} = \Delta E_0 + \Delta E_{\rm ZPV} + \Delta H_{\rm T} + \Delta nRT \tag{2}
$$

where ∆*H*_{f,P} and ∆*H*_{f,R} are HOFs of products and reactants at 298 K, respectively. ∆*E*₀ and ∆*E*ZPV are the differences between the total energies at 0 K and between the zero-point vibrational energies of products and reactants, respectively, and Δ*H*_T is the thermal enthalpy correction from 0 to 298 K. The ΔH_1 s of the reference compounds except CH₃N₃ and $NH₂NO₂$ are taken from the experimental values. For $CH₃N₃$ and $NH₂NO₂$ which lack of experimental data, additional calculations were performed at the G2 level to get accurate ∆*H*^f from the atomization reaction. Previous researches have verified the reliability of the procedure ²³.

Detonation velocity (*D*) and detonation pressure (*P*) were predicted using the empirical Kamlet-Jacobs equations ²⁴ .

$$
D=1.01(N \ \overline{M}^{0.5} \mathcal{Q}^{0.5})^{0.5} (1+1.30\rho)
$$
 (3)

$$
P=1.558 \, \rho^2 N \, \overline{M}^{0.5} \mathcal{Q}^{0.5} \tag{4}
$$

where *ρ* is the density (g.cm⁻³), *N* is the moles of gas produced by per gram of reactant, \overline{M} is the average molar weight of detonation products, and *Q* is the chemical energy of detonation (cal.g⁻¹). N, \overline{M} and Q were determined according to the largest exothermicity principle ²⁵.

Specific impulse (I_s) is the most vital parameter for characterizing and evaluating the performance of propellants. It was calculated using the following equations. First, the proportional specific impulse $I_{s,p}$ was estimated using Eq.(5)²⁶, which has been employed by many studies $15, 27$.

$$
I_{s,p} \sim T_c^{1/2} N^{1/2} \tag{5}
$$

To use Eq. (5), it is necessary to determine the combustion temperature (T_C) and the molar number of products (N) produced by per unit weight of propellant. T_C was calculated using Eq. (6).

$$
\Delta H_{\rm C} = C_{\rm p} (T_{\rm C} - T_0) \tag{6}
$$

where ΔH_C is the enthalpy of combustion and can be calculated from the HOFs of the reactant (ΔH_{fR}) and product (ΔH_{fP}).

$$
\Delta H_{\rm C} = \sum \Delta H_{\rm f, R} - \sum \Delta H_{\rm f, P} \tag{7}
$$

The HOFs of products are known and that of reactant was computed from the above mentioned isodesmic reaction.

Page 5 of 15 RSC Advances

Since Eq. (5) is a proportion relationship, the obtained $I_{s,p}$ isn't an absolute value. So the relative specific impulse $(I_{s,r})$ to HMX $(1, 3, 5, 7$ -tetranitro-1, 3, 5, 7-tetraazacyclooctane) was predicted with Eq. (8) . Then the absolute specific impulse (I_s) of the title compounds was estimated by multiplying the experimental I_s of HMX (265.10 s²⁸) with Eq. (9).

$$
I_{\rm s,r} = I_{\rm s,p} / I_{\rm s,p(HMX)}
$$
\n⁽⁸⁾

$$
I_{\rm s} = 265.10 I_{\rm s,r} \tag{9}
$$

The thermal stability was investigated by evaluating the bond dissociation energy (E_{BD}) . To obtain more accurate results, in this study, E_{BDS} of possible initiation bonds were calculated using Eq. (10).

$$
E_{\rm BD}(A-B)=E_{\rm A}+E_{\rm B}-E_{\rm A-B}-E_{\rm BSSE}
$$
\n(10)

where A-B stands for the neutral molecule and A· and B· for the radical products of the dissociation of the A-B bond; E_{A-B} , E_A and E_B are their corresponding total energies corrected by the zero-point vibrational energies. E_{BSSE} is the basis set superposition error (BSSE) calculated with the method proposed by Boys-Bernardi counterpoise ²⁹, which was usually neglected in pevious works $18, 21-22$.

All calculations except for *D*, *P*, and *I*s were performed with the Gaussian program package³⁰.

3. Results and discussion

3.1 Thermodynamic properties

Thermodynamic properties of the title compounds were evaluated and tabulated in Table 1. It can be seen that $C_{p,m}^0$, S_{m}^0 , and H_{m}^0 increase evidently with the increasing temperature (*T*) from 200 to 800 K. This is because at the higher temperature, the vibrations of molecules are intensified and make more contributions to the thermodynamic properties, but at the lower temperature, the main contributions are from the translation and rotation. Therefore, the increasing temperature leads to the increase of thermodynamic functions.

Table 1 Thermodynamic properties of the title compounds at different temperatures

Property	T/K	$\overline{\mathbf{c}}$ \blacksquare		3	$\overline{4}$	5	6		
$C_{p,m}^0$, J.mol ⁻¹ .K ⁻¹ 200.00 132.32 185.11 241.09					298.40	351.11	404.18	458.81	513.45
	300.00	170.96	243.64	317.27	392.18	464.92	537.40	610.25	683.63
	400.00	205.44	295.64	385.94	477.07	567.57	657.45	747.18	837.77
	500.00	234.00	338.55	442.93	547.83	652.75	757.00	860.94	965.81

Taking compound 4 as an example, the relationships between the thermodynamic functions and the temperature are expressed in Fig. 2(a) and shown as follows (R^2 is the correlation coefficient and *SD* is the standard deviation). Obviously, the increments for $C_{p,m}^0$ and S^0 _{*m*} decrease, while that for H^0 _{*m*} increases with the increasing temperature. The corresponding correlation coefficients are all close to 1. These equations will be helpful for further studies on the other physical, chemical, and explosive properties of the title compounds at various temperatures.

In addition, the thermodynamic properties increase with the increasing number of -CH2NNO2- group under the same temperature, and there are good linear relationships between $C_{p,m}^0$, S_{m}^0 , H_{m}^0 and *n*. As an example, relationships between *n* and thermodynamic properties at 298 K are depicted in Fig. 2(b) and shown as follows:

Page 7 of 15 RSC Advances

Fig. 2 Relationships between $C_{p,m}^0$ (J.mol⁻¹.K⁻¹), S_{m}^0 (J.mol⁻¹.K⁻¹), H_{m}^0 (kJ.mol⁻¹) and $T(K)$ for compound 4 (a), and *n* at 298 K (b)

3.2 Energetic properties

Table 2 lists the molecular weight, oxygen balance $(OB= [16 \times (c-2a-b/2)] \times 100/M)$, where a, b and c are respectively the number of C, H, and O atoms of the compound $C_aH_bO_cN_d$), calculated density, HOF, detonation pressure, and detonation velocity. For comparison, the calculated and experimental detonation properties of HMX are also summarized in Table 2. Evidently, the calculated detonation properties basically agree with the available experimental values. For example, the predicted ρ , *D*, and *P* of HMX are 1.90 g.cm⁻³, 9.08 km.s⁻¹, 37.66 GPa, respectively and the corresponding experimental results are 1.91 g.cm⁻³, 9.10 km.s⁻¹ and 39.00 GPa ³¹; The predicted ρ of 1, 2, and 3 are 1.45, 1.70, and 1.71 g.cm⁻³, respectively and the corresponding experimental results are 1.43, 1.70, and 1.71 g.cm^{-3 32}. The calculated and experimental densities of 4 have a relative big deviation, for which we can not provide a definite reason. One possible reason is that the experiment has provided the density of one of the polymorphs for 4. Comparing the *D* and *P* of 3 and 4 obtained in this work with those of Zhou, et al 33 , a slight deviation exists, which can be attributed to the different fomulas for calculating *D* and *P*. Our results are much closer to those reported in references [3] or [9].

Table 2 Detonation properties of the title compounds

Compound $M/\text{g/mol}^{-1}$ $OB/\%$ $\rho/\text{g.cm}^{-3}$			ΔH_d /kJ.mol ⁻¹ D /km.s ⁻¹		P/GPa
	172.11	-37.19 1.45 (1.43 ³²)	651.04	7.65	22.59

Fig. 3 presents the relationships between ρ , ΔH_f , P , D , OB and n . Obviously, all the parameters increase with *n*, suggesting that incorporating the $-CH₂NNO₂$ - group has a positive effect on the energy performance of aliphatic azido nitramines. There exists a good linear relationship between HOF and *n*, one -CH₂NNO₂ group increases the HOF by 67.59 kJ.mol⁻¹ on average. Compared with the energetic plasticizers widely used in solid propellants, such as ethylene glycol dinitrate (EGDN, ρ =1.48 g.cm⁻³, D=8.00 km.s⁻¹, P=25.07 GPa) and nitroglycerine (NG, ρ =1.54 g.cm⁻³, D=8.03 km.s⁻¹, P= 5.92 GPa)²⁷, all title compounds have better detonation performance and may be potential energetic plasticizers. When *n*=8, *OB*, *D*, and *P* $(-23.18\%, 8.97 \text{ km.s}^{-1})$, and 35.74 GPa, respectively) are very close to the values of HMX. Since using compound 3 as a high-energy oxidant to replace RDX in the modified double-base propellants partly can markedly enhance the burning rate, reduce the pressure exponent, and improve the performance level 10 , the higher *OBs* of 4~8 make them more attractive.

Fig. 3 Relationships between OB , ρ , D , P , ΔH_f and *n*

For evaluating the I_s using Eq. (5) , it is necessary to establish the identities and amounts of gaseous products. In this paper, two kinds of idealized stoichiometric decomposition

Page 9 of 15 RSC Advances

reactions (I and II in Table 3) were designed. Reaction (I) is obtained according to the largest exothermicity principle ²⁵, i.e., all the N atoms turn into N_2 , the O atoms react with the H atoms to give H₂O at first, and then form $CO₂$ with the C atoms. If the number of O atoms is more than that needed to oxidize H and C atoms, the redundant O atoms will convert into O_2 . If the content of O is not enough to fully oxidize H and C atoms, the remaining H atoms will convert into H2, and C atoms will exist as solid-state C. Reaction (II) is obtained according to the principle proposed by Politzer, et al 26 , i.e., O atoms form H₂O with the H atoms first and then CO with the C atoms, rather than $CO₂$ to achieve the maximum of gas products. For the limitation of the quantity of O , the C atoms do not completely convert into $CO₂$, nor CO , that means the true I_s may be between the values obtained from the reactions (I) and (II). Table 3 gives ∆*H*C, *C*p, *T*C, *N*, *I*s,p, *I*s and corresponding reactions. Fig. 4 shows the variation of *I*s with *n*. Obviously, the designed reactions have a significant effect on *I*ss. *I*ss obtained from (II) are higher than those obtained from (I). However, the variation trends are consistent, that is, *I*^s varies up and down with the changes of *n* from 1 to 8 and it reaches the maximum value at *n*=4, and when *n* exceeds 5, *I*^s essentially keeps the constant. It is worth noting that all title compounds possess higher *I*s than HMX, which makes these compounds more attractive.

Fig. 4 Correlations between n and I_s obtained from two kinds of reactions

Fig. 5 depicts the variation of *C*p, ∆*H*C, *T*C, *N* with *n* for Reaction (I). Clearly, there are good linear correlations between C_p , ΔH_c and *n* (Fig. 5(a)). The correlation equations are ∆*H*C=-634.61-505.79*n* (*R*=-0.99991, *SD*=17.78174) and *C*p=130.90+82.87*n* (*R*=0.99899, *SD*=9.85303). *T*_C increases rapidly with *n* from 1 to 2 and changes little when *n* exceeds 4, while *N* decreases continuously with *n* from 1 to 8 (Fig. 5(b)).

Fig. 5 Correlations between C_p , ΔH_c , *N*, T_c and *n* for the reaction (I)

3.3 Stability

3.3.1 Energy gap between frontier orbitals

Based on the frontier orbital theory 34 , the energy gap between the highest occupied molecular orbital (HOMO) and the lowest unoccupied molecular orbital (LUMO) determines the molecular reactivity, such as the ability to absorb light and to react with other species. The molecule with a smaller energy gap is expected to have a higher reactivity and a lower stability in the chemical or photochemical processes with electron transfer or leap 35 . Table 4 summarizes the energies of HOMO and LUMO (E_{HOMO} and E_{LUMO}), and their energy gaps $(E_{\rm g})$. According to the calculated results, the $E_{\rm g}$ s of the title compounds slightly decrease sequentially from 5.55 eV to 5.02 eV with *n* from 1 to 8, which shows that incorporation of the -CH2NNO2- group will slightly decrease the chemical or photochemical stability.

Table 4 Energies of the frontier orbits and their gaps

$E_{HOMO}/a.u.$	-0.27675	-0.27860	-0.28111	-0.27502	-0.28227	-0.27649	-0.27547	-0.27452
$E_{\text{LIMO}}/a.u.$	-0.07293	-0.08014	-0.08332	-0.08728	-0.09506	-0.09000	-0.08996	-0.08993
$E_{\rm g}$ /eV	5.55	5.40	5.38	5. LL	5.09	5.07	5.05	5.02

3.3.2 Stabilization interaction energy

The stabilization interaction energy (E_2) between the donor (*i*) and acceptor (*j*) was estimated based on the NBO analysis using the following equation 36 .

$$
E_2 = q_i F(i,j)^2 / (E_j - E_i)
$$
\n(11)

where q_i is the donor orbital occupancy; E_i and E_j are diagonal elements (orbital energies) and $F(i,j)$ is the off-diagonal NBO Fock matrix element. To elucidate the effect of the number of $-CH_2NNO_2$ - groups on the stability of the N-N bond in $-NNO_2$ groups, the delocalization from the lone pair of oxygen (n_O) into σ^* antibonding orbital of N-NO₂ (σ^*_{N-N}) was studied. The corresponding E_2 of n_O→σ^{*}_{N-N} are summarized in Table 5. The lowest E_2 s of n_O→σ^{*}_{N-N} in each compound are 48.75 , 48.60 , 47.74 , 47.58 , 45.76 , 45.02 , 44.68 , and 43.88 kcal.mol⁻¹, respectively. Since E_2 reflects the delocalization trend of electrons from i to j , and the lower the E_2 , the weaker the interaction 3^7 , we can find that the delocalization trend of electrons from n_0 to σ^*_{N-N} decreases with *n* from 1 to 8. In addition, the lower E_2 corresponds to the longer r_{N-N} . For example, compound 8 has the lowest E_2 of $n_0 \rightarrow \sigma^*_{N-N(4)}$ (see Fig. 1 for the numbering of N-N bonds) and the longest $r_{N-N(4)}$ (1.401 Å), while compound 1 possesses the highest E_2 of n_O→σ^{*}_{N-N(1)} and the shortest $r_{N-N(1)}$ (1.376 Å). Generally, the larger the bond length is, the lower stability the bond has. This suggests that the stability of the title compounds may slightly decrease with the increasing number of $-CH_2NNO_2$ - groups.

				4		6		8
$n_0 \rightarrow \pi^*_{N-N(1)}$	48.75	48.60	47.74	50.84	50.62	46.64	44.88	46.72
$n_0 \rightarrow \pi^*_{N-N(2)}$		48.64	48.62	47.58	54.3	49.38	46.94	48.54
$n_0 \rightarrow \pi^*_{N-N(3)}$					45.76	45.02	52.06	47.62
$n_0 \rightarrow \pi^*_{N-N(4)}$							44.68	43.88

Table 5 Stabilization interaction energy $(E_2/\text{kcal/mol}^{-1})$ of $n_0 \rightarrow \sigma_{N-N}^*$

3.3.3 Bond dissociation energy

The bond dissociation energy (E_{BD}) is often used to investigate the thermal stability and pyrolysis mechanism for energetic compounds. Generally, the bond that requires the minimum energy to break is the weakest bond and is most likely to be the trigger bond. Our previous research ¹¹ on the pyrolysis mechanism of compound 3 showed that the rupture of N-NO2 and N-N2 (via the hydrogen transfer) may happen simultaneously as the initial step of pyrolysis. Since the basis set superposition error has not been considered for the calculation of E_{BD} , E_{BD} (N-NO₂) is overestimated. E_{BD} (N-NO₂) with and without correction of BSSE are 135.62 and 154.83 kJ.mol^{-1 11}, respectively. According to the BSSE-corrected E_{BD} , the pyrolysis may proceed much easier. The BSSE-corrected E_{BD} s of N-NO₂ for all title compounds were listed in Table 6. Clearly, when $n \leq 3$, the pyrolysis is initiated from the rupture of the outside N-N bond, i.e., N-N(1), while when *n*>3, it starts from the breaking of the inner N-N bond. The lowest $E_{BD}(N-NO_2)$ decreases from compound 1 to compound 8, suggesting that incorporating more $-CH_2NNO_2$ - groups into the molecule decrease the stability of the compound. It is noteworthy that all $E_{BD}(N-NO_2)$ of the title compounds satisfy the stability requirement of energetic compounds $(E_{BD} \approx 80 \sim 120 \text{ kJ} \cdot \text{mol}^{-1}{}^{-38})$. To show the variation of the chemical and thermal stability more intuitively, Fig. 6 was plotted. Obviously, the variation trends of E_g , E_{BD} , and E_2 are the same, i.e., they all decrease with the increase of *n*.

Compared with the synthesized $1-4$, compounds 5, 6, 7, and 8 all have good energetic properties, which suggests that they are attractive candidates as the components of propellants. In fact, to be a good energetic additive, it needs to have not only good energetic properties but also good compatibility with other components in the propellants. Since the main purpose of this work is to screen new azido nitramines from the viewpoint of energetic properties and

stabilities and to investigate the role of $-CH₂NNO₂-$ groups in the design of the azido nitramines, the compatibility of the screened compounds with binders (such as GAP) will be investigated in the coming works. We will also investigate whether they are helpful for improving the mechanical properties of propellants.

		2	3	4	\sim	6		8
$N-N(1)$	152.36	140.35	135.62	138.73	135.63	134.84	135.25	144.55
$N-N(2)$		148.27	145.05	132.08	130.56	128.32	139.26	134.95
$N-N(3)$					126.34	122.43	126.68	125.70
$N-N(4)$							119.87	118.83

Table 6 Bond dissociation energy $(E_{BD}/kJ/mol^{-1})$ of N-NO₂

Fig. 6 Variation of $E_{\rm g}$, $E_{\rm 2}$, and $E_{\rm BD}$ with *n*

4. Conclusions

Eight aliphatic azido nitramines were studied using the B3LYP/6-31G* method of density functional theory. Their thermodynamic properties, density, detonation pressure, and detonation velocity increase while stability decreases with the increasing number of $-CH_2NNO_2$ - groups. Specific impulse (I_5) varies up and down, and it reaches the maximum value when $n=4$ and maintains essentially a constant value when $n>5$. All title compounds satisfy the stability requirement of energetic compounds. Compared with the synthesized compounds 1~4, compounds 5~8 are all potential promising plasticizers for propellants. This paper demonstrates how the $-CH_2NNO_2$ - group affects the performance of azido nitramines and provides guidance for the molecular design for this kind of compounds.

Acknowledgments

RSC Advances **Page 14 of 15**

We gratefully thank the National Natural Science Foundation of China (NO. 21403110), the Research Fund for Natural Science Foundation of Jiangsu Province (NO. BK20130755), and the "Excellent Plan and Zijin Star" Research Foundation of NUST for their support to this work. J. Yang would like to thank the Innovation Project for Postgraduates in Universities of Jiangsu Province (grant no. KYLX_0346) for partial financial support.

Reference

- 1. J. E. Flanagan, M. B. Frankel and E. F. *US Pat.*, 4141910, 1979.
- 2. J. E. Flanagan and M. B. Frankel, *US Pat*., 4085123, 1978.
- 3. R. A. Henry and W. P. Norris, *US Pat.*, 4362583, 1982.
- 4. A. Helmy, *Investigation of New Energetic Ingredients for Minimum Signature Propellants,* AIAA/SAE/ASME 20th Joint Propulsion Conference, Cincinnati, Ohio, 1984.
- 5. T. M. Klapoetke, B. Krumm and F. X. Steemann, *Propellants, Explosives, Pyrotechnics*, 2009, **34**, 13-23.
- 6. J. L. Wang, Y. P. Ji, F. L. Gao, W. Guo and S. T. Ren, *Chinese Journal of Energetic Materials*, 2011, **19**, 693-696.
- 7. Y. P. Ji, Y. Lan, P. R. Li, W. Wang, F. Ding and Y. J. Liu, *Chinese Journal of Explosives & Propellants*, 2008, **31**, 44-46.
- 8. Y. Oyumi, A. Rheingold and T. Brill, *Journal of Physical Chemistry* 1987, **91**, 920-925.
- 9. Z. Z. Zhang, B. Z. Wang, Z. C. Shi, Y. P. Ji, Q. Liu and C. H. Zhu, *Chinese Journal of Explosives & Propellants*, 2003, **26**, 3-5.
- 10. Z. M. Dang, F. Q. Zhao, L. S. Wen, C. M. Yin and D. M. Tu, *Journal of Xi'an JiaoTong University*, 2000, **34**, 88-92.
- 11. J. Q. Yang, F. Wang, J. Y. Zhang, G. X. Wang and X. D. Gong, *Journal of Molecular Modeling*, 2013, **19**, 5367-5376.
- 12. C. Lee, W. Yang and R. G. Parr, *Physical Review B*, 1988, **37**, 785-789.
- 13. A. D. Becke, *The Journal of Chemical Physics*, 1993, **98**, 5648-5652.
- 14. P. C. Hariharan and J. A. Pople, *Theoretica Chimica Acta*, 1973, **28**, 213-222.
- 15. G. X. Wang, X. D. Gong, Y. Liu, H. C. Du, X. J. Xu and H. M. Xiao, *Journal of Computational Chemistry*, 2011, **32**, 943-952.
- 16. G. X. Wang, X. D. Gong, Y. Liu, H. C. Du, X. J. Xu and H. M. Xiao, *Journal of Hazardous Materials*, 2010, **177**, 703-710.
- 17. J. Q. Yang, G. X. Wang, F. Wang and X. D. Gong, *Molecular Simulation*, 2014, **40**, 491-497.
- 18. J. Q. Yang, H. Yan, X. L. Zhang, G. X. Wang and X. D. Gong, *Structural Chemistry*, 2014, **25**, 931-940.
- 19. W. M. Haynes, *CRC Handbook of Chemistry and Physics*, CRC press: Taylor and Francis Group LLC, 2012.
- 20. Y. Liu, X. D. Gong, L. J. Wang, G. X. Wang and H. M. Xiao, *The Journal of Physical Chemistry A*, 2011, **115**, 1754-1762.
- 21. F. Wang, H. Du, J. Zhang and X. Gong, *The Journal of Physical Chemistry A*, 2011, **115**, 11788-11795.

Page 15 of 15 RSC Advances

- 22. J. Y. Zhang, H. C. Du, F. Wang, X. D. Gong and Y. S. Huang, *The Journal of Physical Chemistry A*, 2011, **115**, 6617-6621.
- 23. F. Wang, H. C. Du, J. Y. Zhang and X. D. Gong, *The Journal of Physical Chemistry A*, 2011, **115**, 11852-11860.
- 24. M. J. Kamlet and S. Jacobs, *The Journal of Chemical Physics*, 1968, **48**, 23-35.
- 25. P. Politzer and J. S. Murray, *Central European Journal of Energetic Materials*, 2011, **8**, 209-220.
- 26. P. Politzer, J. Murray, M. Grice, P. Sjoberg, G. Olah and D. Squire, *Chemistry of Energetic Materials*, New York, 1991.
- 27. G. X. Wang, X. D. Gong, H. C. Du, Y. Liu and H. M. Xiao, *The Journal of Physical Chemistry A*, 2011, **115**, 795-804.
- 28. Y. F. Liu, H. M. An and R. J. Yang, *Chinese Journal of Explosives & Propellants*, 2001, **24**, 48-49.
- 29. S. F. Boys and F. Bernardi, *Molecular Physics*, 1970, 553-556.
- 30. G. W. T. M. J. Frisch, H. B. Schlegel,, M. A. R. G. E. Scuseria, J. R. Cheeseman, G. Scalmani, V., B. M. Barone, G. A. Petersson, H. Nakatsuji, M. Caricato, X., H. P. H. Li, A. F. Izmaylov, J. Bloino, G. Zheng, J. L. Sonnenberg,, M. E. M. Hada, K. Toyota, R. Fukuda, J. Hasegawa,, T. N. M. Ishida, Y. Honda, O. Kitao, H. Nakai, T. Vreven,, J. J. A. Montgomery, J. E. Peralta, F. Ogliaro, M. Bearpark, J. J., E. B. Heyd, K. N. Kudin, V. N. Staroverov, R. Kobayashi, J., K. R. Normand, A. Rendell, J. C. Burant, S. S. Iyengar,, M. C. J. Tomasi, N. Rega, J. M. Millam, M. Klene, J. E. Knox,, V. B. J. B. Cross, C. Adamo, J. Jaramillo, R. Gomperts, R. E., O. Y. Stratmann, A. J. Austin, R. Cammi, C. Pomelli, J. W., R. L. M. Ochterski, K. Morokuma, V. G. Zakrzewski, G. A. and P. S. Voth, J. J. Dannenberg, S. Dapprich, A. D. Daniels, O. Farkas, J. B. Foresman, J. V. Ortiz, J. Cioslowski, D. J. Fox,, *Gaussian 09, revision A. 02; Gaussian*, 2009.
- 31. M. Talawar, R. Sivabalan, T. Mukundan, H. Muthurajan, A. Sikder, B. Gandhe and A. S. Rao, *Journal of Hazardous Materials*, 2009, **161**, 589-607.
- 32. D. Badgujar, M. Talawar, S. Asthana and P. Mahulikar, *Journal of Hazardous Materials*, 2008, **151**, 289-305.
- 33. X. Zhou and Y. Yu, *Science and Technology of Energetic Materials*, 1991, **52**, 251-259.
- 34. W. J. Hehre, L. Radom, P. v. R. Schleyer and J. A. Pople, *Ab Initio Molecular Orbital Theory*, Wiley, New York 1986.
- 35. H. Dong and F. Zhou, *Performance of High Energetic Explosive and Related Compounds*, Science Press, Beijing, 1989.
- 36. F. Weinhold, in *Natural Bond Orbital Analysis Programs: NBO 5.0/5.G Program Manual*, Madison, Wisconsin, 1996, pp. 1-23.
- 37. W. Zheng, N.-B. Wong, W.-K. Li and A. Tian, *The Journal of Physical Chemistry A*, 2004, **108**, 11721-11727.
- 38. H. M. Xiao, X. J. Xu and L. Qiu, *Theoretical Design of High Energy Density Materials*, Science Press, Beijing, 2008.